

# SLO-based quick notes

## Chapter: 06

### "FLUID DYNAMICS"

Mechanics • The branch of physics dealing with action of forces on matter.

↓  
FLUID Mechanics • The study of fluids either in motion or at rest

↙  
FLUID statics

- study of fluids at rest

↘  
FLUID dynamics

- study of fluids in motion (liquids and gases)

→ FLUID:

The state of matter which can flow from one place to another.

## 1. Viscous Fluids

→ Viscosity: The force of friction b/w different layers of a fluid is called viscosity.

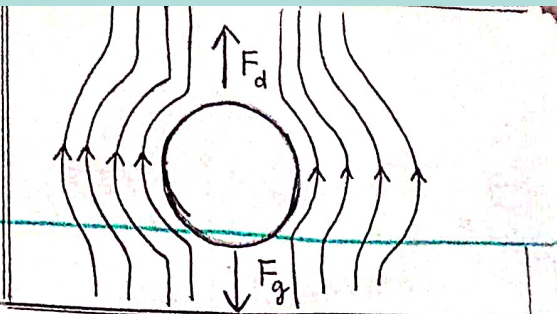
- examples: Honey has high viscosity  
Freely Flowing Gasoline has Low viscosity.
- Intermolecular forces: The stronger the intermolecular forces, the more the viscosity.
- Depends upon: 1) Nature of fluid 2) Cohesive forces  
3) Temperature of fluid

→ Coefficient of viscosity: The numeric value of resistance to flow of a fluid (viscosity) is called coefficient of viscosity. • represented by:  $\eta$

- SI Unit: Pascal second (Pa s)
- Other units:  $\text{kg m}^{-1} \text{s}^{-1}$  or  $\text{Nm}^2 \text{s}$  or dyne second/cm<sup>2</sup>
- Dimension:  $[\eta] = [ML^{-1}T^{-1}]$



# Stokes Law



## Drag Force:

→ An object moving through a fluid experiences a retarding force called drag force.

## Explanation:

- When an object moves through a fluid, the fluid exerts a retarding force that reduces the speed of the object.
- The moving body exerts a force on the fluid to push it out of the way.
- By Newton's 3<sup>rd</sup> law of motion, the fluid pushes back on the body with equal and opposite force.

## Example:

Putting our hand out the window of a fast moving car.

## Factors drag force depends on:

1. Size, shape and orientation
2. Properties of the fluid (viscosity & density)
3. Speed of the object relative to fluid

## Stokes Law:

The viscous drag force on a spherical object is expressed by a formula, termed as Stokes' law.

## Consideration: According to Stokes' law $F_D$

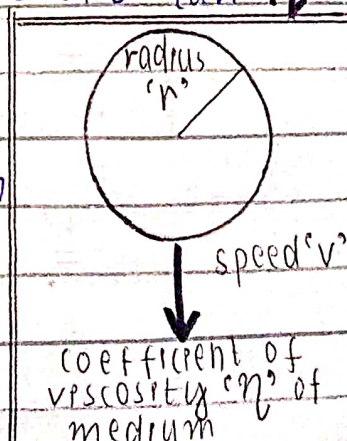
- depends upon 'r' radius of spherical object
- 'v' velocity of spherical object
- 'η' coefficient of viscosity of medium

$$\therefore A = 6\pi$$

$$F_D \propto \eta r v$$

$$F_D = A \eta r v$$

$$F_D = 6\pi \eta r v$$





# 3. Terminal Velocity

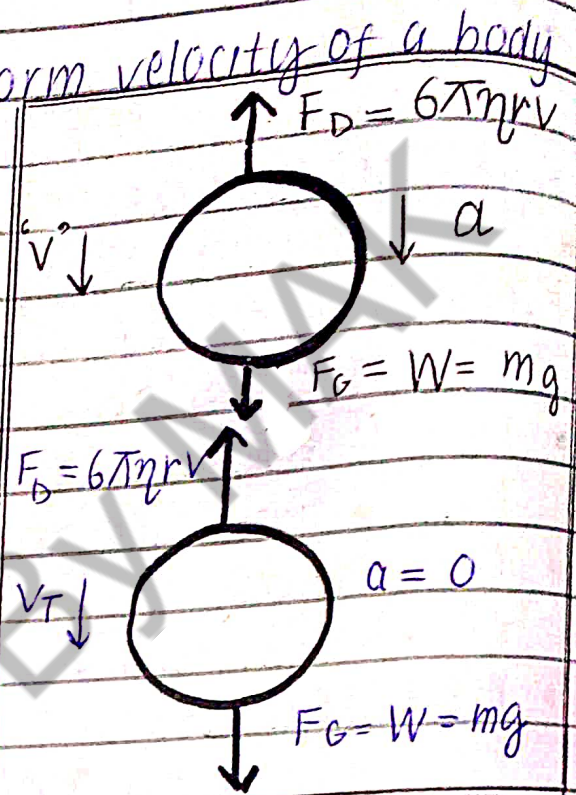
## definition:

- It is the maximum uniform velocity of a body facing fluid friction

## explanation:

- When net force = 0, then acceleration terminates. When this happens, we say that the object has reached its terminal speed

- terminal velocity  $v_T$   $\Rightarrow$  for falling objects  
 $\downarrow$  in downward direction



## Consideration:

- Consider a <sup>spherical</sup> body of radius 'r' moving with 'v' speed through medium of viscosity 'η'. The eq. for net can be written as,

Net force = weight - drag force

$$F = W - F_d$$

$$F_{net} = F_G - F_D$$

$$ma = mg - 6\pi\eta rv$$

$$m(0) = mg - 6\pi\eta r v_T$$

$$6\pi\eta r v_T = mg$$

$$v_T = \frac{mg}{6\pi\eta r} \rightarrow (a)$$

For sphere of uniform density,

$$m = \rho V$$

$$m = \frac{4}{3}\pi r^3 \rho$$

$$V = \frac{4}{3}\pi r^3$$

Putting value of mass in (a)

$$v_T = \frac{4/3\pi r^3 \rho g}{6\pi\eta r}$$

$$v_T = \frac{4\pi r^3 \rho g}{18\pi\eta r}$$

$$v_T = \frac{2\rho g r^2}{9\eta} \rightarrow (b)$$

$$v_T = \text{constant } r^2$$

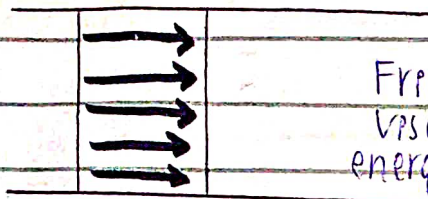
$$v_T \propto r^2$$



# 4. Fluid flow

## → Ideal Fluid Flow:

1. **Non-viscous** - No friction b/w layers of fluid
2. **Steady (Laminar) flow** - velocity of fluid at each point remains constant
3. **Incompressible** - No change in density
4. **Irrrotational** - No angular momentum at any point
5. **Temperature doesn't vary**  $\{ \omega \rightarrow 0 \}$



Friction = 0  
viscosity ( $\eta$ ) = 0  
energy loss = 0

## Laminar Flow

## Turbulent flow

### definition

- |  |  |
|--|--|
| • The flow of a fluid in which every particle moves along a smooth path. | • The irregular/unsteady flow of a fluid is called turbulent flow. |
|--|--|

### Characteristics

- |   |  |
|---|--|
| • Regular, steady or smooth flow of a fluid | • Irregular, unsteady or random flow of a fluid. |
|---|--|

### Effect of velocity

- |                              |                               |
|------------------------------|-------------------------------|
| • It flows with low velocity | • It flows with high velocity |
|------------------------------|-------------------------------|

### Effect of viscosity

- |                         |                        |
|-------------------------|------------------------|
| • It has high viscosity | • It has low viscosity |
|-------------------------|------------------------|

### crossing of fluid layers

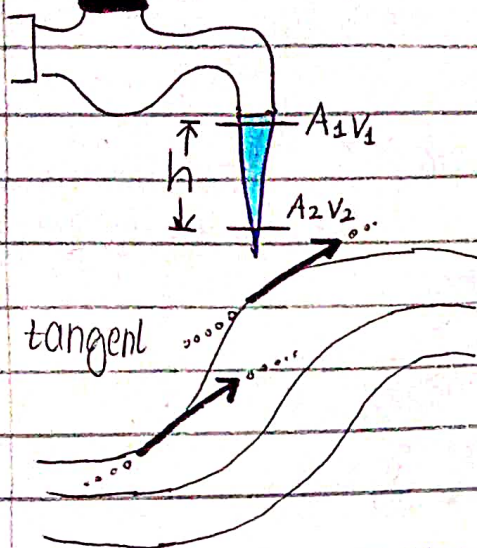
- |  |                                 |
|--|---------------------------------|
| • Fluid layers do not cross each other | • Fluid layers cross each other |
|--|---------------------------------|

### examples

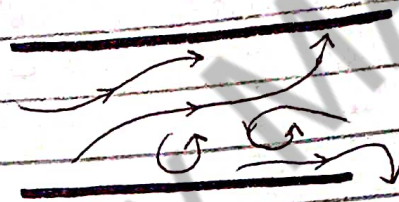
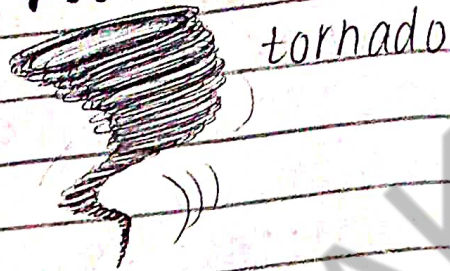
- |                              |   |
|------------------------------|---|
| i) smooth flow of deep water | i) windstorm (of gas particles) <small>turbulent flow</small> |
| ii) flow of gentle breeze    | ii) random motion of smoke particles                          |



## Laminar Flow



## Turbulent Flow



- all points follow the path of others.

### Ideal Fluid

constant and hence called as incompressible

zero

zero

zero

Infinite

→ Laminar flow

→ Spouting can (e.g. of Torricelli's theorem)

### Density

### Viscosity

### Irrotationality

### Surface Tension

### Bulk Modulus

### Real Fluid

compressible, as density varies with applied pressure.

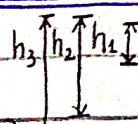
Non-zero

Turbulence in flow exist.

Non-zero

Finite

→ Turbulent flow.



holes for jet

→ weakest jet

→ strongest jet

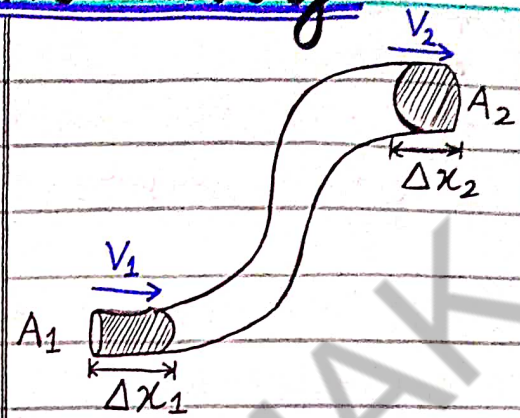


# 5. Equation of Continuity

- Valid only for ideal fluids
- Based on conservation of mass

## → Statement:

The product of cross sectional area and speed of fluid at any point along the pipe is constant



$$A_1 v_1 = A_2 v_2$$

→  $Av = \text{constant}$

## → Proof:

### • consideration:

Consider a pipe of non-uniform size through which an ideal fluid is flowing. Let  $\Delta x_1$ ,  $A_1$  and  $v_1$  are the distance, area and speed of ideal fluid at lower end as shown

∴ rate of flow = volume flown in 1 second

$$\Rightarrow A_1 v_1 = A_2 v_2 \quad \begin{matrix} \therefore v = \text{velocity} \\ \therefore A = \text{area} \end{matrix}$$

$$\Rightarrow Av = \text{constant}$$

$$\Rightarrow v \propto \frac{1}{\text{area}}$$

of pipe above.

### • derivation:

- Distance  $\Delta x_1$  covered by a fluid in time  $\Delta t$  is given by:

$$\Delta x_1 = v_1 \Delta t$$

- Volume contained by a fluid at lower end of pipe is;

$$V = A_1 \Delta x_1$$

$$V = A_1 v_1 \Delta t \quad (\text{value of } \Delta x_1)$$

- Mass contained by ideal fluid at lower end is;

$$\Delta m_1 = V \rho_1$$

$$\Delta m_1 = A_1 v_1 \Delta t \rho_1$$

- Similarly, mass  $\Delta m_2$  at upper-end of pipe

$$\Delta m_2 = A_2 v_2 \Delta t \rho_2$$

- From law of conservation of mass,  $\Delta m_1 = \Delta m_2$

$$A_1 v_1 \Delta t \rho_1 = A_2 v_2 \Delta t \rho_2$$

$$A_1 v_1 = A_2 v_2$$

In general,

$$Av = \text{constant}$$



# 6. Bernoulli's equation

- valid for ideal fluids
- Law of conservation of energy
- involves Pressure, Height and speed

## → Statement:

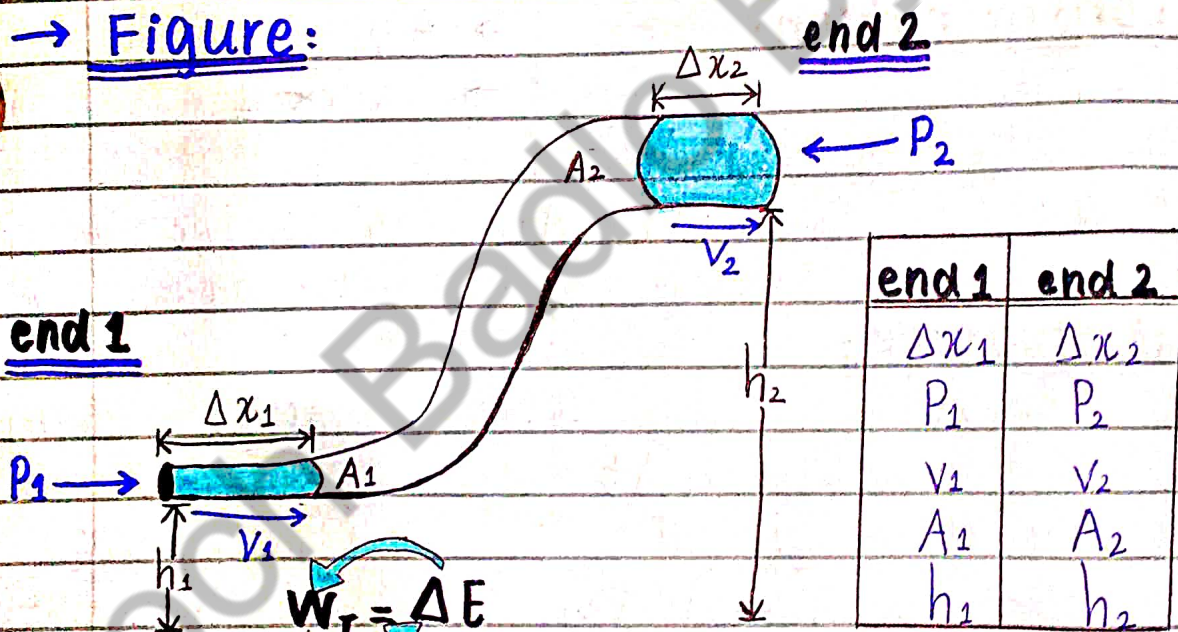
Bernoulli's equation that relates the pressure, flow speed and height of an ideal fluid

Mathematically:

$$P_1 + \frac{1}{2} \rho v_1^2 + \rho g h_1 = P_2 + \frac{1}{2} \rho v_2^2 + \rho g h_2$$

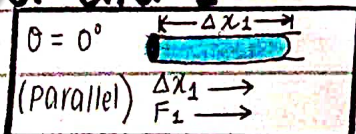
$$P + \frac{1}{2} \rho v^2 + \rho g h = \text{constant}$$

## → Figure:



$W_T = \Delta E$   
 $W_1 + W_2 = \Delta K \cdot E + \Delta P \cdot E$  (finding out)

### → For end-1



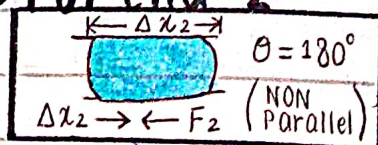
$$W_1 = \vec{F}_1 \cdot \vec{\Delta x}_1$$

$$W_1 = F_1 \Delta x_1 \cos \theta$$

$$W_1 = F_1 \Delta x_1 \cos(0^\circ) \quad \because \cos(0^\circ) = 1$$

$$W_1 = F_1 \Delta x_1$$

### → For end-2



$$W_2 = \vec{F}_2 \cdot \vec{\Delta x}_2$$

$$W_2 = F_2 \Delta x_2 \cos \theta \quad \because \cos(180^\circ) = -1$$

$$W_2 = F_2 \Delta x_2 \cos(180^\circ)$$

$$W_2 = -F_2 \Delta x_2$$



# 1. Pressure

As,  $F = PA$   
 $F_1 = P_1 A_1$  and  $F_2 = P_2 A_2$

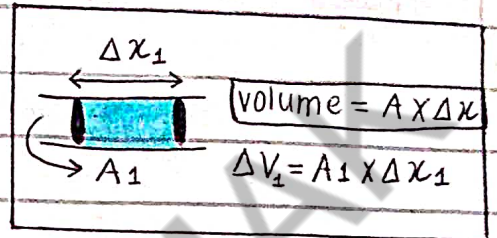
$$\Rightarrow W_1 = F_1 \Delta x_1 \quad \Rightarrow W_2 = -F_2 \Delta x_2$$

$$W_1 = P_1 A_1 \Delta x_1 \quad W_2 = -P_2 A_2 \Delta x_2$$

# 2. Volume (remains same)

As,  $\Delta V_1 = A_1 \Delta x_1$  so,  $W_1 = P_1 \Delta V_1$

As,  $\Delta V_2 = A_2 \Delta x_2$  so,  $W_2 = -P_2 \Delta V_2$



# 3. Density (mass and density remains same)

$$\rho = \frac{\Delta m}{V} \Rightarrow \Delta V = \frac{\Delta m}{\rho}$$

As,  $\Delta V_1 = \Delta V_2 = \Delta V_3$ ; then  $\Delta V_1 = \frac{\Delta m}{\rho}$  and  $\Delta V_2 = \frac{\Delta m}{\rho}$

As,  $W_1 = P_1 \Delta V_1$       As,  $W_2 = -P_2 \Delta V_2$

$$W_1 = \frac{P_1 \Delta m}{\rho}$$

$$W_2 = -\frac{P_2 \Delta m}{\rho}$$

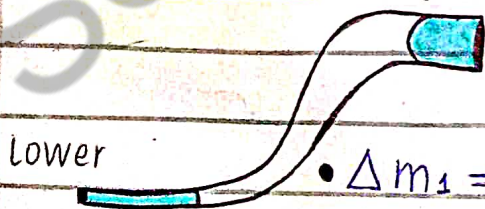
As,  $W_T = W_1 + W_2$

$$W_1 + W_2 = \Delta K \cdot E + \Delta P \cdot E$$

$$\frac{P_1 \Delta m}{\rho} - \frac{P_2 \Delta m}{\rho} = \frac{P_1 \Delta m}{\rho} - \frac{P_2 \Delta m}{\rho}$$

$$W_T = \frac{P_1 \Delta m}{\rho} - \frac{P_2 \Delta m}{\rho}$$

# → ENERGIES:



$$\bullet \Delta K \cdot E = \frac{1}{2} \Delta m_2 v_2^2 - \frac{1}{2} \Delta m_1 v_1^2$$

$$\bullet \Delta U = \Delta m_2 g h_2 - \Delta m_1 g h_1$$

$$\bullet \Delta m_1 = \Delta m_2 = \Delta m$$

$$W = \Delta K \cdot E + \Delta P \cdot E$$

$$\frac{P_1 \Delta m}{\rho} - \frac{P_2 \Delta m}{\rho} = \frac{1}{2} \Delta m_2 v_2^2 - \frac{1}{2} \Delta m_1 v_1^2 + \Delta m_2 g h_2 - \Delta m_1 g h_1$$

as mass is constant  $\Delta m_1 = \Delta m_2 = \Delta m$

$$\frac{P_1 \Delta m}{\rho} - \frac{P_2 \Delta m}{\rho} = \frac{1}{2} \Delta m v_2^2 - \frac{1}{2} \Delta m v_1^2 + \Delta m g h_2 - \Delta m g h_1$$



( $\Delta m$  and  $\rho$  are common)

$$\frac{\Delta m}{\rho} (P_1 - P_2) = \Delta m \left( \frac{1}{2} v_2^2 - \frac{1}{2} v_1^2 + gh_2 - gh_1 \right)$$

$$\frac{1}{\rho} (P_1 - P_2) = \left( \frac{1}{2} v_2^2 - \frac{1}{2} v_1^2 + gh_2 - gh_1 \right)$$

$$P_1 - P_2 = \rho \left( \frac{1}{2} v_2^2 - \frac{1}{2} v_1^2 + gh_2 - gh_1 \right)$$

$$P_1 - P_2 = \frac{1}{2} \rho v_2^2 - \frac{1}{2} \rho v_1^2 + \rho gh_2 - \rho gh_1$$

$$P_1 + \frac{1}{2} \rho v_1^2 + \rho gh_1 = P_2 + \frac{1}{2} \rho v_2^2 + \rho gh_2$$

generally,

$$P + \frac{1}{2} \rho v^2 + \rho gh = \text{constant}$$

## Applications of Bernoulli's eq

### 1. Filter Pumps:

#### ① Definition:

→ A device used to produce partial vacuum in vessel attached to it.

#### ② Components:

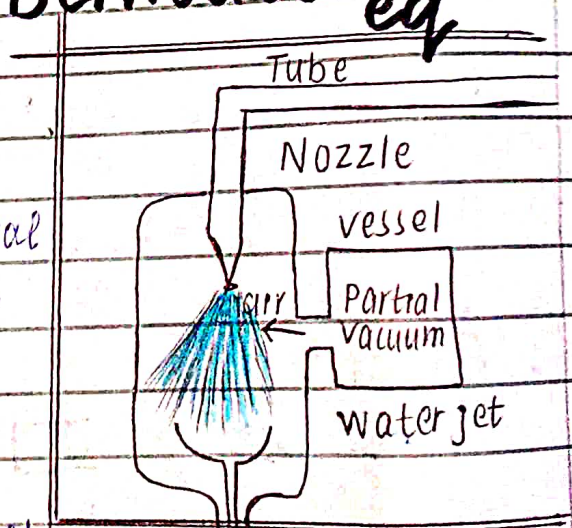
- 1) Tube
- 2) Nozzle
- 3) Water jet
- 4) Vessel

#### ③ Working:

- 1) Water flows from tube towards jet.
- 2) Near jet-section, water's speed increases, pressure drops
- 3) Air flows from the side tube
- 4) Air and water are forced at bottom of filter pump.

#### ④ Usage:

Pumps are used to transfer liquids from low pressure zones → high pressure zones.





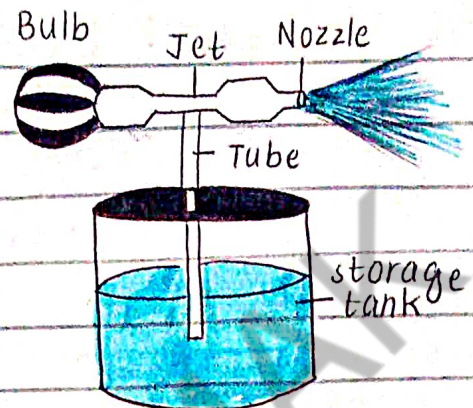
## 2. Atomizers

### → definition:

A device for emitting water, perfume, or other liquids as a fine spray.

### → components:

- 1) Jet
- 2) Nozzle
- 3) Tube
- 4) storage tank
- 5) Bulb



### → Working:

- 1) a stream of air passing over one end of open tube
- 2) The other end is immersed in a liquid, reducing pressure.
- 3) This reduction in pressure, causes the liquid to rise into air stream, ready for dispersion like fine spray.

### → examples: perfume bottles, engine carburetor, water filter pumps and paint sprayers.

## 3. Torcelli's theorem

→ valid for: flowing fluids with negligible viscosity.

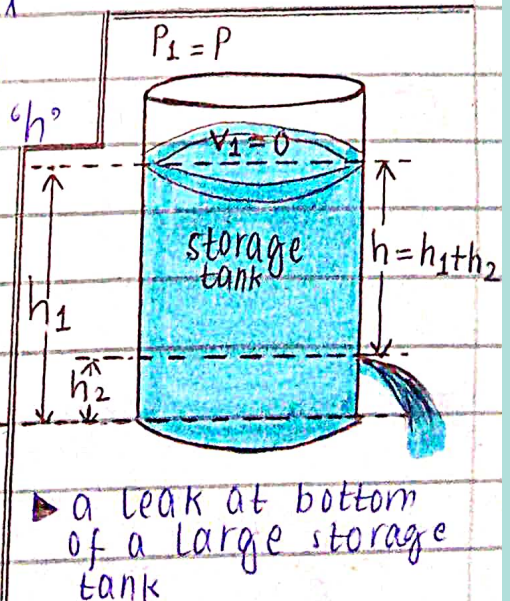
→ Statement: The speed of efflux is equal to the speed gained by fluid while falling through height 'h' under action of gravity.

→ formula:  $v = \sqrt{2gh}$

→ Independent of:  $\rho$  (density)

→ Importance: The Law explains the relation b/w fluid leaving a hole and liquids 'h' at container

→ Example: Spouting can





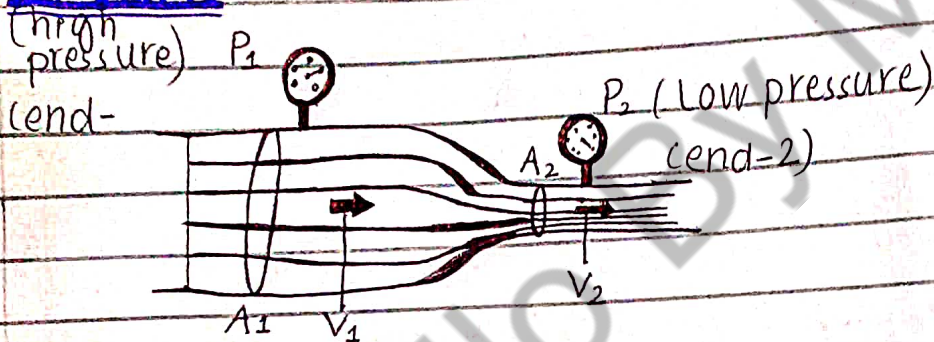
## 4. Venturi Metre (Flow metre)

→ statement/definition:

Venturi metre is a device used to measure the flow speed or flow rate through a piping system

→ Principle: Bernoulli effect.

→ Figure:



→ Flow Velocity:  
mathematically;

$$V_1 = A_2 \sqrt{\frac{2(P_1 - P_2)}{\rho(A_1^2 - A_2^2)}}$$

→ with barometer

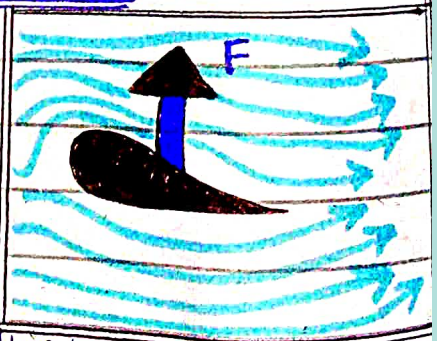
$$V_1 = A_2 \sqrt{\frac{2(gh)}{A_1^2 - A_2^2}}$$

→ without barometer

## 5. Aerofoil

→ definition:

The devices which are shaped so that relative motion b/w it and the fluid produces a force perpendicular to the flow are called aerofoils



→ Shape of aerofoils: stream-lined

→ Aerodynamic force: An airfoil-shaped body moved through a fluid produces an aerodynamic force



→ Principle:  
Bernoulli effect

→ Components:

- perpendicular to direction of motion = lift
- parallel to direction of motion = drag

→ examples:

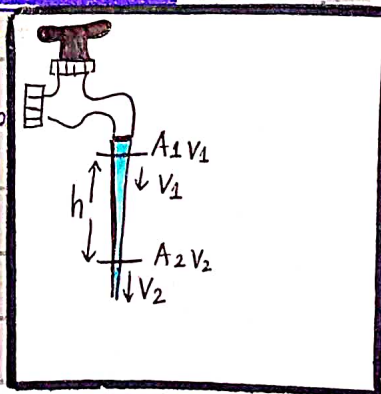
Aeroplane wings, helicopters, sailboats, propellers, fans, compressors etc.



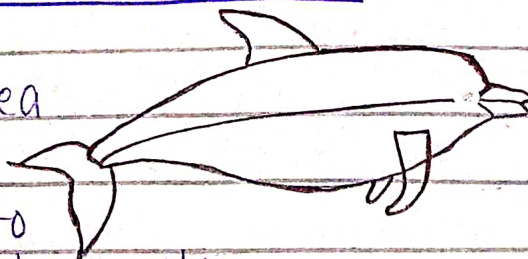
→ The pressure above the wing is reduced relative to the pressure under the wing, as a result is lifted upward.

→ equation of continuity:

- When water falls from a tap, the speed increases under the action of gravity as it falls
- When the speed increases, the <sup>cross</sup> sectional area decreases to keep the equation of continuity valid.



→ Dolphins and other sea-creatures have stream-lined bodies to assist their movement in water.





## 6. Blood Flow

### → Introduction:

- The flow of blood in the arteries is a very good example of Bernoulli's principle.

### → Principle: Bernoulli's effect

### → Dependence:

- The blood flowing in the body depends upon the radius of its arteries

### → Blood velocity and pressure relationship:

- Pressure decreases ; blood velocity increases
- Pressure increases ; blood velocity decreases
- $P \uparrow V \downarrow$  ;  $P \downarrow V \uparrow$

### → Reduction of radius of arteries:

- **arteriosclerosis** results in the **thickening or hardening** of artery walls, thus reducing the radius of arteries

### → Danger of reduction of radius:

It can even result in **heart attacks**

### → Figure: (Master art)

